Comparison of Size and Performance of Small Vertical and Short Takeoff and Landing UAS

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Abstract

This paper presents a data-set of performance characteristics of nearly two hundred vertical/short takeoff and landing (V/STOL) uncrewed aerial systems (UAS). Characteristics of the UAS that are recorded include maximum gross takeoff weight, endurance, maximum length dimension, speed, payload, and payload fraction. The data-set is restricted to small UAS that weigh under 500 lbs. The results are visualized via scatter plots and statistically characterized. The performance of different UAS design types and UAS Group Numbers are compared. The data-set provides a snapshot of current capabilities of small UAS in the V/STOL category and may be useful to UAS developers and procurement agencies.

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TABLE OF CONTENTS

| 1. INTRODUCTION | 1 |
|--------------------------------|----|
| 2. DATA COLLECTION METHODOLOGY | 2 |
| 3. RESULTS AND DISCUSSION | |
| 4. CONCLUSION | 4 |
| REFERENCES | 5 |
| BIOGRAPHY | 10 |
| APPENDICES | 11 |

1. INTRODUCTION

Small uncrewed aerial systems (UAS) with vertical/short takeoff and landing (V/STOL) capabilities are used in applications such as infrastructure inspection, traffic monitoring, public safety, agriculture, aerial videography, medical/package delivery, and in military tasks. Compared to traditional runway-based platforms, V/STOL UAS are more mobile and versatile—they can be deployed with limited infrastructure, in rugged terrain, and in constrained environments (e.g., urban settings, ship decks). Various UAS designs have been developed that enable V/STOL capabilities, including helicopter, quadplane, tiltrotor, tailsitter, multirotor, and fixed-wing designs that are hand or catapult launched. Many V/STOL platforms support hovering and low-speed flight for missions that require holding a fixed or slow-moving position,

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landing/takeoff in constrained environments, or maneuvering at low speed around obstacles.

This study aggregates performance data for V/STOL platforms that weight 500 lbs or less—ranging from miniature first-person-view multirotor for acrobatic flying to larger heavy-lift helicopters. Many commercial platforms exist in this category making it challenging for end-users to decide which designs are most appropriate for their application, or for UAS designers to comprehensively compare new designs to existing systems. The data-set presented here contains nearly two hundred platforms and is analyzed to highlight unique characteristics and trends among different platforms.

Prior Work

Existing literature that reports UAS characteristics includes public vehicle data-sets and academic works. For example, AUVSI's Uncrewed Systems & Robotics Database (USRD) [1] and Janes All The World's Aircraft: Unmanned 22/23 Yearbook [2] provide platform specifications for a broad range of UAS. Open-access data-sets include the Unmanned Systems Technology database [3] and the Center for a New American Security database [4]. Past survey articles of UAS systems include [5], [6]. Work in [7–10] encompasses larger aircraft and discusses empirical correlations between aircraft parameters, such as size, weight, wing and tail geometries, and propulsion. Reference [9] focuses on rotorcraft and suggests empirical scaling laws and characterizes variation in battery characteristics among platforms. The authors in [11, 12] perform similar studies on small fixed-wing, rotarywing, and hybrid UAS that include analysis of datalink range, altitude, endurance, size and weight. Work in [13] discusses propulsion sizing for electric and fuel-powered UAS. Reference [14] provides an overview of the UAS market and compares UAS speed, payload, range, endurance, and propulsion type. In contrast to prior work, this paper focuses specifically on small V/STOL platforms (under 500 lbs) with an emphasis on identifying performance trends among different platform design types.

Contributions

The contributions of this paper are: (1) a data-set of the performance and properties of small vertical/short takeoff and landing uncrewed aerial systems, and (2) a visualization of these data along with their statistical characterization. The resulting data is interpreted and discussed in the context of

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the merits and drawbacks of various V/STOL designs. The overall performance gaps across the range of current V/STOL platforms are also identified. The data-set and analysis of this paper may be useful to UAS developers for preliminary sizing, conceptual design, and performance comparison. It may also benfit procurement agencies to provide a snapshot of the design space occupied by current small-sized UAS in the V/STOL category.

Paper Organization

The remainder of the paper is organized as follows. Section 2 describes the methodology of the data collection process and summarizes the data-set. Section 3 visualizes the data and discusses observed trends and outliers. Section 4 concludes the paper and suggests future work.

2. DATA COLLECTION METHODOLOGY

This section describes the methodology used to identify small V/STOL UAS platforms parameters and summarizes the data-set obtained.

Data Collection

A list of UAS vendors and platform names was first generated from online open-access resources, such as the Blue UAS Cleared List [15], Unmanned Systems Technology database [3], Center for a New American Security database [4], and other published sources [5]. Platforms were investigated further if they met the following criteria: (a) have V/STOL capabilities, (b) were actively in use, and (c) were within the weight class for the study.

Data was collected for UAS with a maximum takeoff weight (MTOW) under 500 lbs. This weight range corresponds to Group 1, Group 2, and a subset of Group 3 UAS classifications as specified by the U.S. Department of Defense (as outlined in Table 1 below). According to other classifications this weight range refers to *small UAS* < 25 kg (< 55 lbs) or a subset of *medium UAS* 25–2,000 kg (55–4,409 lbs) as delineated by the International Telecommunication Union [16]. NATO UAS categorization classifies this weight range as Class I <150 kg (<330 lbs) or a subset of Class II 150–600 kg (330–1323 lbs) with subgroupings of *mini* < 15 kg, *small* < 150 kg, or *tactical* 150–600 kg.

Table 1. U.S. Department of Defense UAS classification [17].This article analyzes Group 1, Group 2, and a subset of Group 3UAS. Altitude values are given in feet above ground level (AGL) or
according to flight level (FL).

| UAS Group | Maximum takeoff weight (lbs) (MTOW) | Nominal operating altitude (ft) | Speed (knots) [mph] |
|-----------|---|---------------------------------------|------------------------|
| Group 1 | 0–20 | < 1,200 AGL | 100 [115] |
| Group 2 | 21–55 | < 3,500 AGL | < 250 [289] |
| Group 3 | < 1,320 | < FL 18,000 | < 250 [289] |
| Group 4 | > 1,320 | < FL 18,000 | Any airspeed |
| Group 5 | > 1,320 | > FL 18,000 | Any airspeed |

Each platform was investigated further by obtaining the corresponding data-sheet or web-page listing its specifications. Other available models from the same vendor were reviewed and included if they met the required criteria. The vendorstated speed (miles per hour), maximum takeoff weight (lbs), payload capacity (lbs), and flight time (minutes) were converted to common units and recorded. UAS size (feet) was recorded as the longest dimension of the vehicle as measured by the platform provider. In the absence of data a particular field was left empty or inferred from other available information (e.g., payload was computed as the difference between the MTOW and empty weight). The payload fraction was computed as the ratio of payload weight to MTOW. The platform type was categorized as either a fixed-wing aircraft, helicopter, multirotor, quadplane/tiltrotor, tailsitter, or firstperson-view (FPV) multirotor drone. Fixed-wing platforms were included and considered to have V/STOL capabilities if they were designed for catapult or hand launch. Loitering munitions and other weaponized UAS were included if all other criteria were met. A UAS Group Number was assigned to each platform by considering the MTOW only (i.e., regardless of speed). The country of origin for each platform was determined based on the address for the headquarters of each vendor. Each entry in the resulting data-set has a known MTOW, length, and speed (entries missing all three parameters were discarded).

Summary of Data-set

In total, the specifications of 189 platforms are included in the presented data-set¹. The distribution of the data across platform types, UAS Group Number, and country of origin is provided in Table 2. The data included platforms from 31 countries. Of these, the five largest countries of origin in the data-set make up 63% of the data. The data represents platforms from 108 unique UAS manufacturers.

| Table 2. | Distribution of data collected among platforn | n |
|----------|---|---|
| types | , UAS group types, and countries of origin | |

| Platform Type | No. Entries | Percentage |
|-----------------------|------------------|------------|
| Fixed-wing | 61 | 33.3 % |
| Helicopter | 26 | 14.2 % |
| Multirotor | 40 | 21.9 % |
| Quadplane / Tiltrotor | 37 | 20.2 % |
| Tailsitter | 11 | 6.0 % |
| FPV Multirotor Drone | 4 | 2.2 % |
| Group 1 | 79 | 43.2 % |
| Group 2 | 59 | 32.2 % |
| Group 3 | 45 | 24.6 % |
| United States | 68 | 36.0 % |
| China | 18 | 9.5 % |
| Israel | 16 | 8.5 % |
| Spain | 9 | 4.8 % |
| Ćanada | 8 | 4.2 % |
| Slovenia | 7 | 3.7 % |
| United Kingdom | 7 | 3.7 % |
| Russia | 6 | 3.2 % |
| Portugal | 6 5 5 | 2.7 % |
| Ukraine | 5 | 2.7 % |
| Netherlands | 4 | 2.1 % |
| Italy | 4 | 2.1 % |
| France | 3 | 1.6 % |
| Germany | 4 3 3 3 | 1.6 % |
| Turkey | 3 | 1.6 % |
| Other | 19 | 10.1 % |

¹The data-set is available online: https://github.com/roboticsuncc/VSTOL_UAS_Database

3. RESULTS AND DISCUSSION

This section presents the data collected in the context of four main platform characteristics: maximum speed, size, payload, and maximum flight time. These characteristics were visualized using the following two techniques:

1. Scatter plots that depict the density of the measured points and illustrate how the characteristics scale with MTOW.

2. Violin plots [18] that cluster UAS either by their design type or UAS Group Number (for UAS Groups 1–3) and depict within-cluster variation. Violin plots (see middle and lower panels in Figs. 1–4) use a shaded area to represent a rotated kernel density estimate of the data. The white circle represents the mean, and the grey bars are inter quartile ranges, similar to a traditional box-and-whisker plot. Data points are jittered along the horizontal axis for clarity.

Size Comparison

The data-set comparing platform size is presented in Fig. 1. The top panel depicts the relationship between UAS maximum takeoff weight and UAS size. As expected, heavier platforms correspond to larger platform dimensions. The relationship between size and weight is positively correlated in the range of 0-100 lbs. Larger platforms weighing over 100 lbs do not exhibit the same correlation. The heaviest platform in the data-set is a large helicopter (Schiebel: Camcopter S-100) which has a weight of 441 lbs and a rotor diameter of 11.2 ft as its maximum dimension. While this vehicle is the heaviest in the data-set, it is nearly half the length of largest platform (by size)-a fixed-wing aircraft with a wingspan of 21 ft and weight of 165 lbs (Aeronautics System: Orbiter 5). Among the different UAS types (middle panel) the FPV multirotors were both the lightest and smallest platforms, whereas quadplanes/tiltrotors and helicopters had the largest mean size.

As expected, size is correlated with increasing UAS Group Number (lower panel). The mean UAS size was approximately 4.6 ft, 7.5 ft and 10.4 ft, for Groups 1, 2, and 3, respectively.

Speed Comparison

Figure 2 (top panel) compares UAS maximum takeoff weight and maximum attainable speed. Most of the platforms studied (89%) have a maximum speed of 140 mph and are less than 140 lbs. Multicopters are among the slowest platforms with the exception of FPV multirotor type models (see middle panel). The distribution of the speed data appears clustered around the mean for most UAS types with the exception of a few outliers. The Group 1 outlier is a highly optimized FPV Multirotor Drone (RacerX) that can reach a speed of 179 mph and weighs only 1.80 lbs. The Group 3 outliers include a turbojet powered quadplane (Woot Tech: Firebolt) and a turbojet powered quadrotor (Wave Aerospace: Sea Huntress II). Other quadplanes use conventional propeller-driven systems. The result highlight the difficulty of designing high-speed UAS (> 140 mph) that are relatively small and lightweight.

A significant proportion of the Group 1 and Group 2 UAS (last panel) have speed either above or below the 115 mph cutoff for Groups 1 and 2 (see Table 1). Similarly, Group 3 UAS have maximum speeds as low as 40 mph (typical for Group 1) but all Group 3 platforms have speeds below the 287 mph speed boundary between Groups 3 and 4. These results illustrate that many UAS cannot be unambiguously classified into a specific UAS Group Number when both weight and speed are considered.

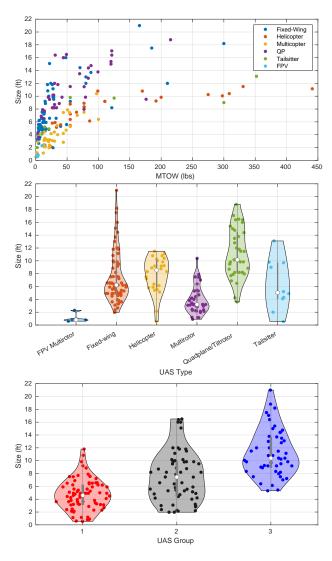


Figure 1. Comparison of UAS size with maximum takeoff weight (MTOW), UAS type, and UAS Group Number.

Flight Time Comparison

Figure 3 (top panel) depicts the correlation between UAS MTOW and the UAS flight time. About 75% of the platforms have a flight endurance of four hours or less and 36% have a flight endurance of 1 hour or less. The data in the fourth panel has a distinctive bottom-heavy skew; many platforms achieve low flight endurance times (e.g., less than 3 hours) and greater endurance values are less common. Fixed-wing, tailsitter, and quadplane/tiltrotor platform types achieve substantially higher flight times than multirotors, FPV drones, or helicopters. Group 1 UAS are generally limited to 5 hours or less of flight time. Group 2 UAS have an intermediate endurance of 12 hours or less and Group 3 UAS have endurances surpassing 24 hours. The outliers in the UAS Group 2 category include a gas-powered quadplane (Tekever: AR3) with an endurance of 18 hours and a gas-powered tailsitter (Volatus Flexrotor) with an endurance of 24 hours. Electric platforms often have lower endurance than their gas-powered counterparts.

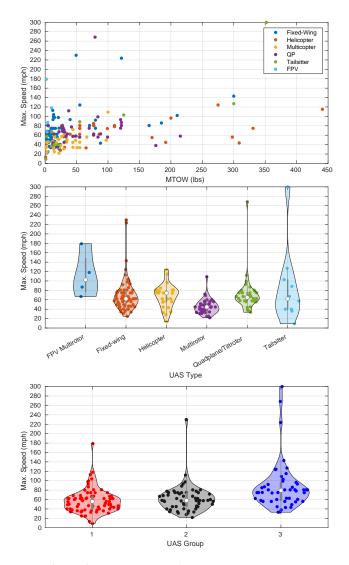


Figure 2. Comparison of UAS maximum speed (miles/hour) with maximum takeoff weight (MTOW), UAS type, and UAS Group Number.

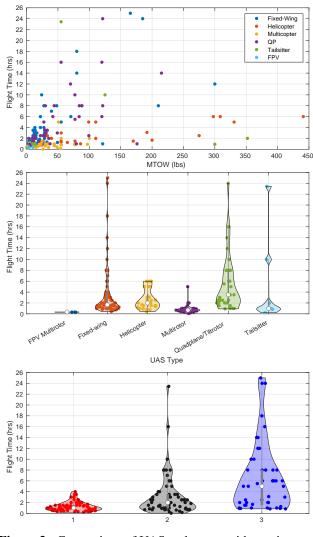


Figure 3. Comparison of UAS endurance with maximum takeoff weight (MTOW), UAS type, and UAS Group Number.

Payload Comparison

Typical payloads onboard V/STOL UAS include imaging sensors (visual/infrared spectrum), hyperspectral cameras, thermal sensors, LiDAR, GPS, acoustic transducers, gas analyzers, and synthetic aperture radar. Helicopters have the largest mean payload capacity and, like tailsitters, can carry payloads up to approximately 110 lbs (see Fig. 4). Fixed wing aircraft and FPV multirotor drones have the smallest payload capacity, whereas multirotors and quadplanes/tiltorotors have intermediate capacity. Group 3 UAS are on average twice as large as Group 1 UAS but can carry over three times as much payload. The payload fraction with maximum takeoff weight (MTOW) is plotted in Fig. 5. The data shows that some smaller platforms (MTOW <100 lbs) have large payload fractions of 0.5 or greater. Heavier platforms generally have smaller payload fractions that decrease with weight.

4. CONCLUSION

A data-set of the characteristics of 189 vertical short takeofflanding (V/STOL) platforms that weigh less than 500 lbs was generated using open-source information. The platforms were characterized according to their size, maximum speed, flight endurance, and payload capabilities. The variation in these characteristics was then visualized against (a) maximum takeoff weight, (b) UAS platform types (fixed-wing, helicopter, multirotor, quadplane/tailsitter, FPV multirotor drone), and (c) UAS Group Number (either Group 1, 2, or 3). The analysis illustrates how the characteristics scale as platform size increases and the variation among different UAS designs. Outliers were identified and their design features were discussed. Future studies may consider enlarging the data-set and recording additional characteristics (e.g., ceiling altitude, battery characteristics, and communication/control range), and developing empirical relations among these parameters.

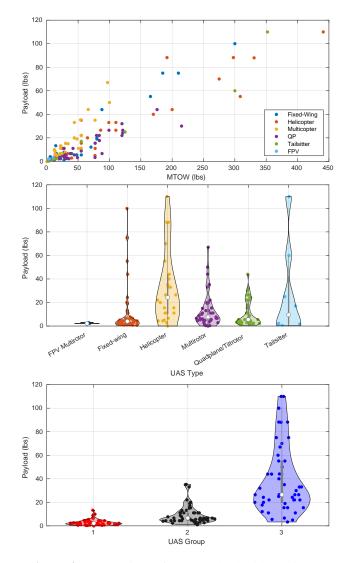


Figure 4. Comparison of UAS payload with maximum takeoff weight (MTOW), UAS type, and UAS Group Number.

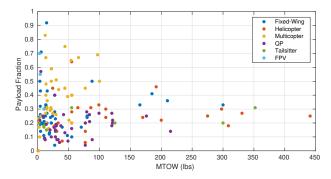


Figure 5. Comparison of UAS payload fraction with maximum takeoff weight (MTOW).

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APPENDICES

The data-set is listed in the following three tables. Some vendor and platform model names appear truncated due to space constraints (details can be inferred from the corresponding reference cited).

| No. | Туре | Vendor | Model | Weigh (lbs) | tSize (ft) | Speed (mph) | | Pay -load frac. | Endur (hours | UAS)Group | Country |
|----------------|----------------|------------------------------------|--|----------------|---------------|----------------|----------------|-----------------------|-----------------|--|----------------------------|
| 1 | HELI | VELOS | V3 [19] | 55.0 | 6.4 | 74.6 | 35.0 | 0.64 | 1.33 | 2 | USA |
| 2 | HELI | Aerovironment | VAPOR [20] | 65.0 | 7.5 | 33.0 | 20.0 | 0.31 | 1.25 | 3 | USA |
| 3 | HELI | UAVOS | UVH 25EL [21] | 55.0 | 8.8 | 62.1 | 11.0 | 0.20 | 1.50 | 2 | USA |
| 4 | HELI | Steadicopter | Black Eagle 50H [22] | 110.2 | 9.2 | 78.3 | 26.5 | 0.24 | 5.00 | 3 | Israel |
| 5 | HELI | Anduril | Ghost [23, 24] | 37.0 | 8.9 | 85.0 | 10.0 | 0.27 | 0.92 | $\begin{vmatrix} 2 \\ 2 \end{vmatrix}$ | USA |
| 6 7 | HELI | UAVOS | UVH 170 [25] | 99.0 | 10.8 | 74.0 | 33.0 | 0.33 | 5.00 | 3 | USA Switzerland |
| 8 | HELI HELI | Aeroscout UMS Skeldar | Scout B1-100 [26] V-150 [27] | 170.0 330.7 | 10.8 | 55.0 74.6 | 40.0 88.0 | 0.24 | 1.50 5.00 | 33 | Switzerland Switzerland |
| 9 | HELI | ZIYAN UAS | Falcon-10 [28] | 55.1 | 5.4 | 80.8 | 15.4 | 0.27 | 0.83 | 3 | China |
| 10 | HELI | ZIYAN UAS | Blowfish A2G [29] | 86.0 | 6.1 | 62.1 | 26.5 | 0.31 | 1.00 | 3 | China |
| 11 | HELI | ZIYAN UAS | Ranger P2X PTK [30] | 39.7 | 5.5 | 62.1 | 2.9 | 0.07 | 2.00 | 2 | China |
| 12 | HELI | Alpha Unmanned | Alpha 800 [31,32] | 30.9 | 5.9 | 34.2 | 4.4 | 0.14 | 2.50 | 2 | Spain |
| 13 | HELI | Swiss Drones | SDO 50 V2 [33] | 191.8 | 9.2 | 44.7 | 88.2 | 0.46 | 3.10 | 3 | Switzerland |
| 14 | HELI | Schiebel | Camcopter S-100 [34] | 441.0 | 11.2 | 115.1 | 110.0 | 0.25 | 6.00 | 3 | Austria |
| 15 | HELI | Steadicopter | Black Eage 50E [22, 35] | 110.2 | 9.2 | 80.5 | 33.1 | 0.30 | 2.00 | 3 | Israel |
| 16 | HELI | Steadicopter | Black Eage 50 [22, 36] | 77.2 | 8.3 | 80.5 | 11.0 | 0.14 | - 12 | 3 | Israel |
| 17 | HELI | FLIR | Black Hornet [37] | 0.7 | 0.6 | 13.4 | 0.0 | 0.00 | 0.42 | 1 | USA |
| 18 19 | HELI | 4Front Robotics | Navig8 Electric [38] | 24.2 200.0 | 2.1 9.8 | 28.0 96.3 | $11.0 \\ 44.0$ | 0.45 | 0.75 | $ ^{2}_{3}$ | Canada |
| 20 | HELI HELI | 4Front Robotics ZALA | Navig8-32 Gas [39,40] 421-02 [41] | 200.0 | 10.1 | 55.9 | 44.0 88.2 | 0.22 | 1.67 6.00 | 3 | Canada Russia |
| 20 | HELI | ZALA | 421-06 [41, 42] | 297.0 | 5.8 | 80.5 | 4.4 | 0.30 | 1.50 | $\frac{3}{2}$ | Russia |
| 22 | HELI | Indela | I.N. Sky [43] | 308.6 | 10.4 | 43.5 | 55.1 | 0.18 | 6.00 | 3 | Belarus |
| 23 | HELI | Drone Hopper | Nuntius [44] | 33.1 | 5.2 | 55.9 | 6.6 | 0.20 | 2.50 | 2 | Spain |
| 24 | HELI | Drone Hopper | Titanium [44] | 55.1 | 7.2 | 74.6 | 22.1 | 0.40 | 3.00 | 3 | Spain |
| 25 | HELI | Steadicopter | Black Eagle 35E [22] | 77.2 | 8.3 | 80.5 | 15.4 | 0.20 | 1.50 | 3 | Israel |
| 26 | HELI | 4Front Robotics | Navig8-56 Gas [40] | 275.0 | 10.2 | 124.3 | 70.0 | 0.25 | 2.50 | 3 | Canada |
| 27 | FPV | DJI | FPV [45] | 4.0 | 0.8 | 87.0 | 2.2 | 0.55 | 0.33 | 1 | China |
| 28 | FPV | DRL | RacerX [46] | 1.8 | 0.8 | 179.0 | - | - | - | 1 | USA |
| 29 | FPV | Lumenier | QAV-PRO [47] | 10.0 | 2.3 | 118.0 | 3.0 | 0.30 | 0.33 | 1 | USA |
| 30 | FPV | DJI Dormot | Avata [48] | 3.1 | 0.6 | 67.0 | 2.2 | 0.70 | 0.30 | 1 | China |
| 31 32 | MULTR MULTR | Parrot | ANAFI USA [49] Skydio 2+ [50] | 1.4 1.7 | 1.2 | 33.0 36.0 | 0.3 0.0 | 0.23 | 0.53 0.45 | 1 | France USA |
| 33 | | Skydio | Skydio X2 [50] | 3.2 | 2.9 | 25.0 | 0.0 | 0.00 | 0.45 | 1 | USA |
| 34 | | DJI | Mavic 3 [51] | 12.1 | 1.1 | 47.0 | 10.0 | 0.83 | 0.72 | 1 | China |
| 35 | | Height Technolo | MI-3 [52] | 24.5 | 4.2 | 38.0 | 6.6 | 0.27 | 1.50 | 2 | Netherlands |
| 36 | | Lockheed Martin | Indago 3 [53] | 5.0 | 3.8 | 46.0 | 1.3 | 0.25 | 1.17 | 1 | USA |
| 37 | MULTR | Teledyne FLIR | Skyranger R70 [54] | 29.7 | 4.4 | 31.0 | 7.7 | 0.26 | 0.98 | 2 | USA |
| 38 | | Ascent Technolo | Spirit [55] | 13.5 | 2.1 | 60.0 | 6.5 | 0.48 | 0.53 | 1 | USA |
| 39 | MULTR | | Kargu [56] | 17.9 | 3.2 | 44.0 | 2.9 | 0.16 | 0.50 | 1 | Turkey |
| 40 | | Inspired Flight | IF800 [57] | 16.9 | 4.4 | 56.0 | 6.6 | 0.39 | - | 1 | USA |
| 41 | | Acecore Technol | Zoe Zetona 8 [58] | 26.4 | 3.1 | 57.0 | 4.8 | 0.18 | 0.03 | $ ^{2}_{2}$ | Netherlands |
| 42 43 | | Ascent Technolo Performance Dro | NX30 [59] C100 [60] | 30.0 34.0 | 3.0 5.4 | 60.0 45.0 | 15.3 15.0 | 0.51 | $0.62 \\ 1.00$ | $\binom{2}{2}$ | USA USA |
| 44 | MULTR | | Sherpa [61] | 35.0 | 4.0 | 50.0 | 15.0 | 0.44 | 0.33 | $\left \begin{array}{c} 2 \\ 2 \end{array} \right $ | USA |
| 45 | | Wave Aerospace | Falcon II LE [62] | 44.0 | 4.2 | 72.0 | 20.0 | 0.45 | 1.00 | $ \frac{2}{2}$ | USA |
| 46 | | Draganfly | Commander 3 XL [63] | 55.0 | 5.3 | 45.0 | 22.0 | 0.40 | 0.83 | $\overline{2}$ | Canada |
| 47 | | Wave Aerospace | X-5B Huntress [64] | 100.0 | 6.4 | 109.0 | 50.0 | 0.50 | 2.00 | 3 | USA |
| 48 | MULTR | Free Fly System | Alta X [65] | 77.0 | 7.5 | 60.0 | 35.0 | 0.45 | 0.83 | 3 | USA |
| 49 | MULTR | | Matrice 350 RTK [66] | 20.3 | 2.9 | 51.5 | 6.0 | 0.29 | 0.92 | 2 | China |
| 50 | MULTR | | Matrice 30 [67] | 22.1 | 2.2 | 51.5 | 13.1 | 0.59 | 0.68 | 2 | China |
| 51 | MULTR | DJI | Mavic 2 Enterpr [68] | 2.4 | 1.2 | 44.7 | 0.4 | 0.17 | 0.52 | 1 | China |
| 52 53 | MULTR | Draganfly | Airspeak S1 [69] Heavy Lift Dron [70] | 15.4 97.0 | 2.1 | 49.2 | 5.5 67.0 | 0.36 | 0.37 0.91 | $\begin{vmatrix} 1 \\ 3 \end{vmatrix}$ | USA Canada |
| 55 54 | MULTR | Sky Drones | X700 [71] | 13.2 | 2.3 | 37.3 | 8.8 | 0.69 | 1.00 | 1 | UK |
| 55 | MULTR | Aerialtronics | Altura Zenith A [72] | 21.3 | 2.0 | 35.7 | 6.6 | 0.31 | 0.67 | 2 | UK |
| 56 | | Sky Drones | Full Throttle A [73] | 22.4 | 2.2 | 34.5 | 13.2 | 0.59 | 0.72 | $\overline{2}$ | UK |
| 57 | | Indro Robotics | Wayfinder [74] | 44.1 | 7.5 | 34.2 | 33.1 | 0.75 | 0.83 | $\overline{2}$ | Canada |
| 58 | MULTR | Indro Robotics | Endurance [75] | 22.1 | 2.4 | 21.8 | 11.0 | 0.50 | 0.67 | 2 | Canada |
| 59 | | Harris Aerial | Carrier H6 [76] | 50.7 | 7.9 | 33.6 | 11.0 | 0.22 | 5.00 | 2 | USA |
| 60 | | Harris Aerial | Carrier H6 Hydr [77] | 55.1 | 5.3 | 33.6 | 11.0 | 0.20 | 2.00 | 3 | USA |
| 61 | | Inspired Flight | IF750 [78] | 15.0 | 3.2 | 40.0 | 4.6 | 0.31 | 0.62 | 1 | USA |
| 62 | | Inspired Flight | IF 1200 [79] | 48.5 | 4.7 | 49.0 | 19.0 | 0.39 | 0.40 | 2 | USA |
| 63 64 | MULIK | BlueHalo | Intense Eye 2 [80] Belugadrone [81] | 13.0 | 2.5 | 40.0 | 3.5 | 0.27 | 0.73 | $\frac{1}{2}$ | USA |
| 64 65 | MULTR | Eurolink System | Belugadrone [81] X6v2 [82] | 22.1 14.6 | 3.2 3.4 | 69.3 25.0 | 6.6 4.4 | 0.30 | 1.00 0.33 | $ ^{2}_{1}$ | Italy Italy |
| 05 | | Drone Hopper | X-Quad [83] | 66.2 | 7.0 | 55.9 | 4.4 | 0.50 | 0.55 | $\begin{bmatrix} 1\\3 \end{bmatrix}$ | Spain |
| 66 | | L'IONC INCOPPOI | 12x Quuu 1001 | | | | | | 0.17 | | |
| 66 67 | | | DH-Agro Hopper1 [83] | 54.0 | 7.4 | 55.9 | 35.3 | 10.65 | 0.17 | 12 | Spain |
| 66 67 68 | MULTR | Drone Hopper T-Drones | DH-Agro Hopper1 [83] MX860 [84] | 54.0 44.0 | 7.4 | 55.9 44.7 | 35.3 19.8 | 0.65 | 0.17 0.67 | $ ^{2}_{2}$ | Spain China |
| 67 | MULTR MULTR | Drone Hopper | | | | | | | | | |

Table 3. Data collected for helicopter (HELI), first-person-view multirotor drone (FPV), and multirotor (MULTR) platforms.

| No. | Туре | Vendor | Model | Weigh | | Speed | | Pay | Endur | | Country |
|-----|-------|-----------------|-------------------------|-------|------|-------|----------------|----------------|--------|------------------|-------------|
| | | | | (lbs) | (ft) | (mph) | -load (lbs) | -load frac. | (nours |)Group | |
| 71 | QP/TR | Krossblade | Prowler [87] | 5.1 | 3.6 | 80.0 | 2.9 | 0.57 | 0.92 | 1 | USA |
| 72 | QP/TR | Quantum Systems | Trinity F90 [88] | 11.0 | 7.9 | 38.0 | 1.0 | 0.09 | 1.50 | 1 | Germany |
| 73 | ÕP/TR | Deltaquad | Deltaquad Pro [89] | 13.7 | 7.7 | 62.6 | 2.6 | 0.19 | 2.00 | 1 | Netherlands |
| 74 | QP/TR | C-Astral | SQA eVTOL [90] | 22.0 | 9.5 | 67.0 | 2.2 | 0.10 | 2.50 | 2 | Slovenia |
| 75 | QP/TR | CUAV | Raefly VT260 [91] | 30.0 | 8.7 | 67.0 | 5.5 | 0.18 | 3.50 | 2 2 | China |
| 76 | QP/TR | Tekever | A3 [92] | 55.0 | 11.5 | 56.0 | 8.8 | 0.16 | 16.00 | 2 | Portugal |
| 77 | ÕP/TR | Aurora Flight S | SKIRONX [93] | 49.0 | 16.5 | 58.2 | 3.2 | 0.07 | 3.00 | 2 | USA |
| 78 | ÒP/TR | CUAV | Raefly VT370 [94] | 77.0 | 8.0 | 60.4 | 3.3 | 0.04 | 10.00 | 3 | China |
| 79 | ÒP/TR | JOUAV | CW-30E [95] | 84.0 | 14.4 | 56.0 | 17.7 | 0.21 | 8.00 | 3 | China |
| 80 | QP/TR | L3 Harris | FVR-90 [96] | 120.0 | 15.4 | 75.0 | 32.0 | 0.27 | 16.00 | 3 | USA |
| 81 | ÕP/TR | Woot Tech | Alien X VTOL [97] | 120.0 | 15.0 | 93.0 | 22.0 | 0.18 | 6.00 | 3 | USA |
| 82 | ÒP/TR | Edge Autonomy | VXE30 [98] | 44.0 | 16.0 | 57.5 | 5.5 | 0.13 | 8.00 | 2 | USA |
| 83 | QP/TR | Edge Autonomy | Penguin C Mk 2. [99] | 70.0 | 13.5 | 74.8 | - | - | 12.00 | 2 3 | USA |
| 84 | QP/TR | Ukrspec Systems | PD-2 [100] | 121.3 | 16.4 | 87.0 | 24.3 | 0.20 | 8.00 | 3 | Ukraine |
| 85 | ÕP/TR | Ukrspec Systems | Leleka 100 [101] | 12.1 | 6.5 | 43.5 | 1.0 | 0.08 | 2.50 | 1 | Ukraine |
| 86 | ÒP/TR | Sky Drones | SkyLane-250 [102] | 33.1 | 8.2 | 58.2 | 2.6 | 0.08 | 3.50 | 2 | UK |
| 87 | ÒP/TR | Sky Drones | SkyLane-350 [102] | 77.2 | 11.5 | 62.6 | 15.4 | 0.20 | 5.50 | 2 3 | UK |
| 88 | QP/TR | Sky Drones | Action Drone [103, 104] | 26.5 | 8.2 | 78.3 | 4.4 | 0.17 | 2.50 | 2 | UK |
| 89 | QP/TR | Elevonx | Skyeye Sierra [105] | 27.6 | 10.2 | 68.3 | 11.0 | 0.40 | 5.00 | 2 2 2 3 | Slovenia |
| 90 | QP/TR | Elevonx | Tango VTOL [105] | 41.9 | 9.8 | 77.7 | 11.0 | 0.26 | 6.00 | 2 | Slovenia |
| 91 | QP/TR | Hammerhead | eV20 [106] | 176.0 | 9.5 | 38.0 | 44.0 | 0.25 | 1.00 | 3 | USA |
| 92 | QP/TR | Woot Tech | Firebolt [107] | 79.4 | 9.8 | 268.4 | 22.1 | 0.28 | 1.00 | 3 | USA |
| 93 | QP/TR | Woot Tech | Firefly [108] | 37.5 | 8.9 | 111.8 | 6.6 | 0.18 | 1.00 | 2 1 | USA |
| 94 | QP/TR | Tekever | A4 [109, 110] | 8.8 | 6.9 | 33.5 | 2.2 | 0.25 | 2.00 | 1 | Portugal |
| 95 | QP/TR | Carbonix | Domani [111, 112] | 88.2 | 14.8 | 62.6 | 6.6 | 0.08 | 8.00 | 3 | Australia |
| 96 | QP/TR | Carbonix | Volanti [113] | 35.3 | 11.8 | 62.6 | 2.2 | 0.06 | 2.00 | 23 | Australia |
| 97 | QP/TR | Aeronautics | Trojan [114] | 99.2 | 13.8 | 55.9 | 26.5 | 0.27 | 2.50 | 3 | Israel |
| 98 | QP/TR | BlueBird | WanderB-VTOL [115] | 28.7 | 10.2 | 74.8 | 3.0 | 0.10 | 2.50 | | Israel |
| 99 | QP/TR | Threod Systems | EOS VTOL UAS [116] | 31.1 | 16.4 | 66.8 | 2.4 | 0.08 | 3.00 | 2 2 3 | Estonia |
| 100 | QP/TR | Threod Systems | Stream C VTOL [117] | 83.8 | 12.8 | 99.0 | 22.1 | 0.26 | 6.00 | 3 | Estonia |
| 101 | ÕP/TR | Soko Aerial | ARACE ROC [118] | 29.8 | 8.2 | 71.6 | 5.5 | 0.19 | 3.50 | 2 | Ghana |
| 102 | ÒP/TR | Aerovironment | Jump 20 [119] | 215.0 | 18.8 | 58.0 | 30.0 | 0.14 | 14.00 | 3 | USA |
| 103 | QP/TR | Event 38 Unmann | E400 [120] | 20.0 | 9.8 | 35.7 | 3.0 | 0.15 | 1.50 | 1 | USA |
| 104 | QP/TR | T-Drones | VA25 [121] | 28.7 | 8.2 | 74.6 | 4.4 | 0.15 | 3.50 | 2 | China |
| 105 | QP/TR | Flight Wave | Edge 130 Blue [122] | 3.4 | 4.2 | 65.0 | 0.8 | 0.22 | 2.00 | 1 | USA |
| 106 | QP/TR | Aeronautics | Orbiter 4 [123] | 121.3 | 17.1 | 80.6 | 26.5 | 0.22 | 24.00 | 3 | Israel |
| 107 | QP/TR | Lockheed Martin | Stalker XE [124] | 30.0 | 12.0 | 44.9 | 5.5 | 0.18 | | 2 | USA |
| | - | 1 | | 1 | | ' | | i. | 1 | | |

 Table 4. Data collected for quadplane/tiltrotor (QP/TR) and tailsitter (TAIL) platforms.

| No. | Туре | Vendor | Model | Weigh (lbs) | tSize (ft) | Speed (mph) | | Pay -load frac. | Endur (hours | UAS)Group | Country |
|------------|--------------|----------------------------|---------------------------------------|----------------|---------------|----------------|--------------|-----------------------|-----------------|---------------|------------------|
| 118 | FWNG | Aerovironment | Raven B RQ-11 [125] | 4.4 | 4.5 | 50.0 | 1.9 | 0.43 | 1.25 | 1 | USA |
| 119 | FWNG | C-Astral | Bramor ppX [126] | 10.4 | 7.5 | 49.0 | 2.2 | 0.21 | 3.50 | 1 | Slovenia |
| 120 | FWNG | Aerovironment | Wasp III [127] | 14.4 | 2.4 | 40.0 | 13.3 | 0.92 | 0.75 | 1 | USA |
| 121 | FWNG | Aerovironment | Puma 3 AE [128] | 15.0 | 9.2 | 47.0 | 4.0 | 0.27 | 2.50 | 1 | USA |
| 122 | FWNG | Black Swift | S2 [129] | 20.8 | 10.0 | 40.0 | 5.0 | 0.24 | 1.83 | 2 | USA |
| 123 | FWNG | Aeronautics | Orbiter 1K [130] | 28.7 | 9.5 | 57.0 | 6.6 | 0.23 | 2.00 | 2 | Israel |
| 124 | FWNG | Textron | Aerosonde Mk 4. [131] | 80.0 | 12.0 | 74.8 | 20.0 | 0.25 | 14.00 | 3 | USA |
| 125 | FWNG | Boeing Insitu | ScanEagle 3 [132] | 80.0 | 13.0 | 92.0 | 19.0 | 0.24 | 18.00 | 3 | USA |
| 126 | FWNG | Aerovironment | T-20 [133] | 185.0 | 17.5 | 86.0 | 75.0 | 0.41 | 24.00 | 3 | USA |
| 127 | FWNG | Northrup Grumma | Bat UAS [134] | 210.0 | 12.0 | 102.0 | 75.0 | 0.36 | 8.00 | 3 | USA |
| 128 | FWNG | Resolute ISR | Resolute Eagle [135] | 300.0 | 18.2 | 143.0 | 100.0 | 0.33 | 12.00 | 3 | USA |
| 129 | FWNG | Ukrspec Systems | Shark UAS [136] | 27.6 | 11.2 | 80.8 | 1.0 | 0.04 | 4.00 | 2 | Ukraine |
| | FWNG | Ukrspec Systems | Mini Shark [137] | 11.0 | 6.6 | 74.6 | 1.0 | 0.09 | 2.00 | 1 | Ukraine |
| 131 | FWNG | ElevonX | Skyeye Delta [138] | 13.8 | 7.5 | 62.1 | 4.4 | 0.32 | 3.50 | 1 | Slovenia |
| 132 | FWNG | ElevonX | Skyeye Sierra [138] | 27.6 | 9.8 | 77.7 | 11.0 | 0.40 | 8.00 | 2 | Slovenia |
| 132 | FWNG | AgEagle | eBee X [139] | 3.5 | 3.8 | 68.3 | 1.8 | 0.50 | 1.50 | 1 | USA |
| 134 | FWNG | EMT | Aladin [140] | 8.8 | 4.8 | 55.9 | 1.6 | 0.19 | - | 1 | Germany |
| 135 | FWNG | Lockheed Martin | Desert Hawk III [141] | 10.2 | 4.9 | 57.5 | 2.0 | 0.19 | 1.50 | 1 | USA |
| | FWNG | Lockheed Martin | Desert Hawk IN [141] | 10.2 | 4.9 | 63.3 | 2.0 | 0.20 | 2.50 | 1 | USA |
| 137 | FWNG | Lockheed Martin | Desert Hawk EER [141] | 24.0 | 12.0 | 40.3 | 6.0 | 0.20 | 10.00 | 2 | USA |
| 137 | FWNG | EMT | Luna X-2000 [142] | 88.0 | 13.7 | 43.0 | 44.0 | 0.23 | 8.00 | $\frac{2}{3}$ | Germany |
| 139 | FWNG | IDETEC | | 4.2 | 5.1 | 46.6 | 1.3 | 0.30 | 0.75 | 1 | Chile |
| | FWNG | Aeronautics | AG-Wing [143, 144] Orbiter 2 [145] | 28.7 | 9.8 | 57.5 | 4.0 | 0.32 | 3.00 | $\frac{1}{2}$ | Israel |
| | | | Orbiter 2 [145] Orbiter 3 [146] | | | | | | | $\frac{2}{3}$ | |
| 141 142 | FWNG FWNG | Aeronautics Aeronautics | Orbiter 5 [146] Orbiter 5 [147] | 70.6 165.4 | 14.4 21.0 | 80.6 80.6 | 12.1 55.1 | 0.17 | 6.00 25.00 | 3 | Israel Israel |
| | | | | | 9.2 | | | | | | |
| 143 | FWNG | IDS Corporation | IA-17 Manta [148] | 55.1 | 7.2 | 124.3 | 5.5 | 0.10 | 5.00 | 3 | Italy |
| 144 | FWNG | Aircraft Trader | Guaridan Eye [149] | 13.2 | | | 4.4 | 0.33 | 4.00 | | Belgium |
| | FWNG | National Chung- | Cardinal II [150] | 13.0 | 6.2 | 34.0 | 4.0 | 0.31 | - | 1 | Taiwan |
| 146 | FWNG | L3 Harris | Cutlass [151] | 15.0 | 4.6 | 97.8 | 3.0 | 0.20 | 1.00 | 1 | USA |
| 147 | FWNG | Aeroland | AL-4 [152] | 9.3 | 6.6 | 62.0 | 2.2 | 0.24 | 1.00 | 1 | China |
| 148 | FWNG | WB Group | Flyeye mini UAV [153] | 26.5 | 11.8 | 74.6 | 4.4 | 0.17 | 2.50 | 2 | Poland |
| 149 | FWNG | Survey Copter | Tracker 120 [154] | 19.2 | 10.8 | 55.9 | 2.4 | 0.13 | 1.50 | 1 | France |
| 150 | FWNG | Aerofoundry | Watupa-e [155] | 22.1 | 15.1 | 31.1 | 4.4 | 0.20 | 6.00 | 2 | Brazil |
| 151 | FWNG | ZALA | 421-08 [41, 156] | 4.6 | 2.7 | 80.8 | 0.7 | 0.14 | 1.67 | 1 | Russia |
| 152 | FWNG | ZALA | 421-04M [41] | 9.3 | 5.2 | 74.6 | 2.2 | 0.24 | 2.00 | 1 | Russia |
| 153 | FWNG | ZALA | 421-16 [41] | 39.7 | 5.3 | 93.2 | 6.6 | 0.17 | 7.00 | 2 | Russia |
| 154 | FWNG | C-Astral | Bramor C4EYE [157] | 10.4 | 7.5 | 67.1 | 2.2 | 0.21 | 3.50 | 1 | Slovenia |
| 155 | FWNG | FT Sistemas | FT-100 [158] | 15.4 | 8.9 | 38.0 | 6.6 | 0.43 | 2.00 | 1 | Brazil |
| 156 | FWNG | Event 38 Unmann | E384 [159] | 5.6 | 6.2 | 45.0 | 4.0 | 0.71 | 1.50 | 1 | USA |
| 157 | FWNG | Event 38 Unmann | E386 [160] | 5.6 | 6.2 | 45.0 | 1.1 | 0.20 | 1.42 | 1 | USA |
| 158 | FWNG | ZOHD | Talon Rebel [161] | 2.8 | 3.3 | 62.1 | - | - | - | 1 | China |
| 159 | FWNG | EADS Cassidian | Tracker / DRAC [162] | 18.7 | 11.8 | 62.1 | 2.2 | 0.12 | 1.50 | 1 | France |
| 160 | FWNG | Lockheed Martin | Stalker XE [124] | 48.0 | 16.0 | 58.0 | 5.5 | 0.11 | 8.00 | 2 | USA |
| 161 | FWNG | SPE Athlon Avia | A1-S Furia [163] | 12.1 | 6.4 | 62.1 | - | - | 3.00 | 1 | Ukraine |
| 162 | FWNG | STM | Alpagut [164, 165] | 121.3 | 8.2 | 223.7 | 24.3 | 0.20 | 1.00 | 3 | Turkey |
| 163 | FWNG | Tekever | AR1 Blue Ray [166] | 16.4 | 5.9 | 34.2 | - | - | 3.00 | 1 | Portugal |
| 164 | FWNG | Tekever | AR4 Light Ray C [167] | 6.6 | 3.6 | 36.0 | - | - | 0.75 | 1 | Portugal |
| 165 | FWNG | Tekever | AR4 Light Ray E [168] | 6.6 | 3.6 | 49.7 | - | - | 0.75 | 1 | Portugal |
| 166 | FWNG | Baykar | Bayraktar Mini [169] | 32.5 | 6.6 | 46.0 | - | - | 1.33 | 2 | Turkey |
| 167 | FWNG | Blue Bear Syste | Blackstart [170] | 16.4 | 4.9 | 74.6 | - | - | 1.00 | 1 | UK |
| 168 | FWNG | Raytheon | Coyote [171] | 21.0 | 4.8 | 97.6 | - | - | 1.50 | 2 | USA |
| 169 | FWNG | Leonardo Airbor | CREX-B [172] | 6.9 | 5.6 | 68.3 | - | - | 1.25 | 1 | Italy |
| | FWNG | Integrated Dyna | Desert Hawk [173, 174] | 14.8 | 4.9 | 62.1 | - | - | 1.00 | 1 | Spain |
| | FWNG | Israel Aerospac | Green Dragon [175, 176] | 49.2 | 5.6 | 230.0 | - | - | 1.25 | 2 | Israel |
| 172 | FWNG | MicroUAV | HawkMoth [177] | 13.4 | 6.5 | 103.8 | - | - | 2.00 | 1 | USA |
| 173 | FWNG | Sky-Watch | Heidrun V1 [178] | 7.2 | 5.4 | 66.5 | | - | - | 1 | Denmark |
| 174 | FWNG | UVision | HERO-120 [179] | 27.6 | 2.0 | 74.6 | 7.7 | 0.28 | 1.00 | 2 | Israel |
| 175 | FWNG | Irkut Engineeri | Irkut-3 [180] | 9.8 | 6.6 | 55.3 | - | - | 1.25 | 1 | Russia |
| | FWNG | BlueBird Aero S | MicroB [181, 182] | 2.2 | 3.1 | 51.8 | 0.5 | 0.24 | 1.00 | 1 | Israel |
| 177 | FWNG | Innocon | MicroFalcon LP [183] | 13.2 | 5.9 | 74.6 | 4.4 | 0.33 | 2.00 | 1 | Israel |
| 178 | FWNG | Sparkle Tech | Pigeon [184] | 4.2 | 3.9 | 62.1 | 1.1 | 0.26 | 1.50 | 1 | China |
| | FWNG | Integrated Dyna | Pride [185] | 9.9 | 5.0 | 62.1 | 1.1 | 0.11 | 1.00 | 1 | Spain |
| | FWNG | UCONSYSTEM | REMOEYE-002B [186, 187] | 11.2 | 5.9 | 49.7 | - | - | 1.00 | 1 | South Korea |
| 181 | FWNG | Integrated Dyna | Rover Mk I [188] | 6.6 | 4.9 | 62.1 | - | - | 0.75 | 1 | Spain |
| 182 | FWNG | Integrated Dyna | Skycam-W [189] | 5.5 | 3.3 | 60.0 | 1.1 | 0.20 | 1.50 | 1 | Spain |
| | FWNG | Skywalker | Skywalker X6 [190, 191] | 4.4 | 4.9 | 24.9 | - | - | 0.42 | 1 | China |
| 184 | FWNG | Blackbar Engine | STORM [192] | 12.0 | 7.6 | 112.8 | - | - | 1.33 | 1 | USA |
| 185 | FWNG | IPCD | Tactical UAV [193] | 4.4 | 5.6 | 49.7 | _ | - | 0.75 | 1 | Indonesia |
| | FWNG | Lockheed Martin | Vector Hawk [194] | 4.0 | 3.6 | 80.8 | 0.8 | 0.19 | 1.17 | 1 | USA |
| 187 | FWNG | WB Group | Warmate [195, 196] | 12.6 | 4.6 | 93.2 | 3.1 | 0.25 | 0.83 | 1 | Poland |
| | FWNG | Innocon | MicroFalcon LE [197] | 22.1 | 6.6 | 74.8 | 4.4 | 0.20 | | 2 | Israel |
| | | | | • | • | • | | | • | | • |

 Table 5. Data collected for fixed-wing (FWNG) platforms.